

## Fluid Mechanics Qualifying Examination Syllabus

Students planning to take the Fluid Mechanics qualifying examination are expected to demonstrate competency in the following areas of Fluid Mechanics:

1. Differential form of the conservation equations governing fluid flow
  - Kinematics, rate of strain tensor and vorticity
  - Stress tensor and linear momentum equation
  - Constitutive equations and fluid properties (viscosity, surface tension, density, etc.)
  - Mass continuity equation
  - Navier-Stokes equations
2. Incompressible Newtonian flows
  - Classical solutions of the Navier-Stokes equations
  - Laminar and turbulent boundary layers
  - Laminar and turbulent free shear flows
  - Laminar flows in ducts
  - Elements of stability theory
  - Skin friction and drag on streamlined bodies
3. Ideal fluid flows
  - Stream function and velocity potential
  - Circle theorem and method of images
  - Conformal mapping
  - Superposition of sources, sinks, vortices, doublets, and uniform streams
  - Lift and drag on airfoils, Kutta-Jukowski theorem
  - Flows over cylinders, spheres, and bluff bodies
4. Compressible fluid flow
  - Isentropic one-dimensional flow in variable area ducts, nozzles, and diffusers
  - Rankine-Hugoniot jump conditions
  - Stationary normal shocks
  - Fanno flow, Rayleigh flow
  - Oblique shocks, Prandtl-Meyer expansions

It is recommended that before taking the Fluid Mechanics qualifying examination, students take the following courses offered by the MechSE department (or their equivalent at another institution):

- ME 410      Intermediate Gas Dynamics
- ME 411      Viscous Flow and Heat Transfer
- TAM 435     Intermediate Fluid Mechanics

Suggested text books for the above material include:

*Compressible Fluid Flow*: Patrick H. Oosthuizen and William Carscallen, published by McGraw Hill

*Viscous Fluid Flow*: Frank M. White, published by McGraw Hill

*Fundamental Mechanics of Fluids, 3<sup>rd</sup> Edition*: Iain G. Currie, published by Taylor & Francis

**QUALIFYING EXAMINATION  
FOR  
Fluid Mechanics**

Department of Mechanical and Industrial Engineering  
University of Illinois at Urbana-Champaign

Wednesday, August \_\_, \_\_\_\_  
9:00 AM – 12:00 PM

**IMPORTANT EXAMINATION INFORMATION**

1. Identify your examination and work with your University Identification Number (UIN, I-Card number in blue beginning with 65) on each page. **DO NOT ENTER YOUR NAME ANYWHERE IN THE EXAMINATION.**
2. Choose 3 out of the 4 problems.
3. Each problem counts 10 points.
4. Start each problem in a new examination booklet and write on only the right-hand side (front side) of each sheet.
5. Hand in this problem package with your exam booklets.

Subject: FLUID: Question (1)

### COMPRESSIBLE STEADY FLUID FLOW

Consider air flow in a converging channel with an inlet area of  $60 \text{ cm}^2$  and an exit area of  $40 \text{ cm}^2$ . At the inlet the average velocity is  $100 \text{ m/sec}$ , the pressure is  $150 \text{ kPa}$ , and the temperature is  $300 \text{ K}$ . [See schematic below. Note we assume adiabatic flow.]

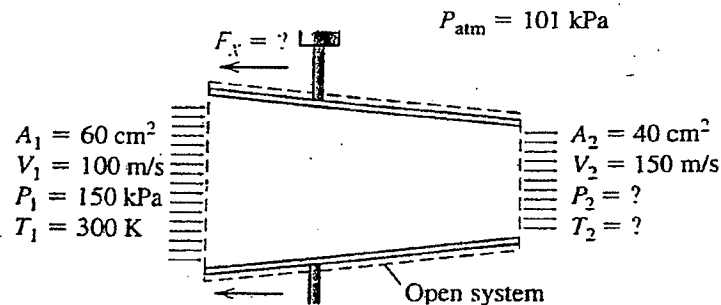
The velocity at the exit is  $250 \text{ m/sec}$ . [We will need to find the exit pressure and the temperature.]

The surrounding atmospheric pressure, is  $101 \text{ kPa}$ . Treat air as an ideal gas, so that the density  $= p/RT$ , and the enthalpy,  $h = c_p \cdot T$ . [Recall for air,  $R = 0.287 \text{ kJ/kg} \cdot \text{K}$  and assume  $c_p = 1.00 \text{ kJ/kg} \cdot \text{K}$ .]

#### PROBLEM:

Using conservation of mass, energy and momentum, show all work and calculate:

- Mass flow rate through the channel (kg/sec);
- The exit pressure (kPa);
- The exit temperature (K);
- The net Force,  $F$  required to keep the channel in place (kN).



Poiseuille Flow: Fully Developed Pipe Flow (Intro)

The velocity distribution for steady, laminar ( $Re_D \ll 2100$ ), fully developed flow through a circular pipe with diameter  $D$  as shown in Figure 7.3 can be determined by using the conservation of mass and linear momentum in cylindrical coordinates  $\mathbf{V} = \mathbf{V}(r, \theta, z)$  (~~given in Appendix C~~), and the same boundary conditions and assumptions used in the plane Poiseuille flow. Additionally, we assume that the flow is axisymmetric (i.e., the flow is independent of the azimuthal direction  $\theta$ ). Therefore, all quantities that depend on  $\theta$  are set to zero. This flow is known as the *Hagen-Poiseuille flow* or simply the *Poiseuille flow*.

The equations of continuity and conservation of linear momentum for the two dimensional laminar flow problem for a Newtonian fluid in the cylindrical coordinate shown in Figure 7.3 are

Governing Equations

Continuity

$$\frac{1}{r} \frac{\partial}{\partial r} (rv_r) + \frac{\partial v_z}{\partial z} = 0.$$

r-momentum

$$\rho \left( v_r \frac{\partial v_r}{\partial r} + v_z \frac{\partial v_r}{\partial z} \right) = \rho g_r - \frac{\partial P}{\partial r} + \mu \left[ \frac{\partial}{\partial r} \left( \frac{1}{r} \frac{\partial}{\partial r} (rv_r) \right) + \frac{\partial^2 v_r}{\partial z^2} \right].$$

z-momentum

$$\rho \left( v_r \frac{\partial v_z}{\partial r} + v_z \frac{\partial v_z}{\partial z} \right) = \rho g_z - \frac{\partial P}{\partial z} + \mu \left[ \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial}{\partial r} (v_z) \right) + \frac{\partial^2 v_z}{\partial z^2} \right].$$

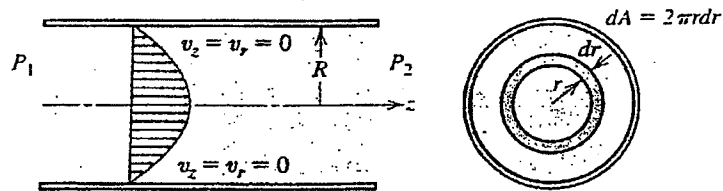


FIGURE 7.3 Fully developed flow in a pipe.

Solution

The fully developed assumption implies that  $v_r = v_r(r)$ ,  $v_z = v_z(r)$ , and that  $\frac{\partial v_r}{\partial z} = \frac{\partial v_z}{\partial z} = 0$ . The continuity equation

$$\frac{1}{r} \frac{\partial}{\partial r} (rv_r) + \frac{\partial v_z}{\partial z} = 0,$$

then reduces to

$$\frac{1}{r} \frac{\partial}{\partial r} (rv_r) = 0 \Rightarrow v_r = \text{constant},$$

(cont'd)

# FLUIDS Question (2): (cont'd)

(P. 4/2)

where, since  $v_r$  cannot be a function of  $z$ , it must be a constant number. Moreover, since  $v_r = 0$  at the pipe wall, the continuity equation is satisfied only when  $v_r$  is zero everywhere in the flow.

Again, using the assumption of steady, fully developed flow with negligible body forces, the momentum equations reduce to

$$0 = -\frac{\partial P}{\partial r}$$

$$0 = -\frac{\partial P}{\partial z} + \mu \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial}{\partial r} (v_z) \right)$$

Upon integration, the momentum equation in the  $r$ -direction gives

$$\frac{\partial P}{\partial r} = 0 \Rightarrow P = P(z),$$

indicating that the pressure is only a function of  $z$ . Consequently,

$$\frac{\partial P}{\partial z} = \frac{dP}{dz}$$

As discussed in Section 7.1, the pressure gradient is always a negative number.

Returning to the momentum in the  $z$ -direction

$$\underbrace{\mu \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial}{\partial r} (v_z) \right)}_{\text{function of } r} = \underbrace{\frac{dP}{dz}}_{\text{function of } z}$$

Since the left-hand side is only a function of  $r$  and the right-hand side is only a function of  $z$ , the two are equal when they are both constants.

By rearranging,

$$\frac{\partial}{\partial r} \left( r \frac{\partial}{\partial r} (v_z) \right) = r \frac{1}{\mu} \frac{dP}{dz}$$

and by integrating

$$\frac{\partial u}{\partial r} = \frac{1}{\mu} \frac{dP}{dz} \frac{r}{2} + \frac{C_1}{r}$$

(A)

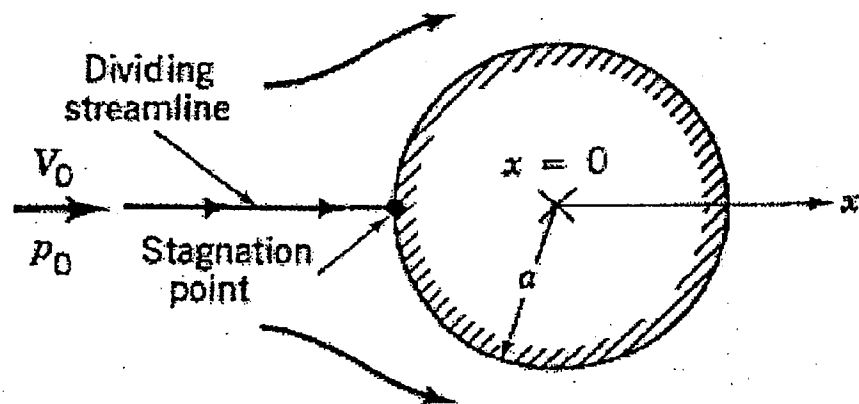
## WHAT'S REQUIRED:

(a) Integrate equation (A) and apply to appropriate Boundary Conditions (showing all work) to solve for the velocity profile as a function of radius and 'other' constant parameters.

(b) Calculate the volumetric flow rate (m<sup>3</sup>/sec),  $\dot{Q} = \int_0^R u(r) dA$

(c) Calculate the pressure gradient,  $dp/dz$ , if  $\dot{Q} = 1 \text{ m}^3/\text{sec}$ ;  $R = 1 \text{ cm}^2$ ; and  $\mu = 1.8 \text{ E-5 (Nt-sec)/m}^2$ .

3. Consider steady, incompressible and inviscid flow past a circular cylinder, as shown in the figure below. The fluid velocity along the dividing streamline ( $-\infty \leq x \leq -a$ ) is given by  $V = V_0 (1 - a^2/x^2)$ , where  $a$  is the radius of the cylinder and  $V_0$  is the upstream velocity.
- Determine the pressure gradient along the dividing streamline. Note that  $\partial P/\partial x = -\rho V \partial V/\partial x$  along that streamline.
  - If the upstream pressure is  $P_0$ , integrate the pressure gradient to obtain the pressure distribution  $P(x)$  for  $-\infty \leq x \leq -a$ .
  - Show from the result of part (b) that the pressure at the stagnation point is  $P(-a) = P_0 + \rho V_0^2/2$ , as expected from the Bernoulli equation.
  - Which flow assumptions (steady, incompressible or inviscid ?) are necessary for the validity of the general Bernoulli equation ?



4. The pump shown in the schematic below adds 25 kW to the water and causes a flow rate of  $0.04 \text{ m}^3/\text{s}$ . Determine the flow rate expected if the pump is removed from the system. The pipe is 30 m in length and of 60 mm internal diameter. The water exits at a 40 mm diameter nozzle. Assume a friction coefficient  $f = 0.016$  for either case (with and without the pump) and neglect minor losses.

